NASA CONTRACTOR REPORT

NASA CR-61358

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# HAZARD ESTIMATES FOR SELECTED ROCKET FUEL COMPONENTS AT KENNEDY SPACE CENTER

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May 5, 1971

Prepared for

NASA-GEORGE C. MARSHALL SPACE FLIGHT CENTER Marshall Space Flight Center, Alabama 35812

TECHNICAL	REPORT	STANDARD	TITLE	PAGE

1. REPORT NO.	2. GOVERNMENT ACC	ESSION NO.	3. RECIPIENT'S CAT	ALOG NO.
NASA CR-61358	<u> </u>			
4. TITLE AND SUBTITLE			5. REPORT DATE	
HAZARD ESTIMATES FOR			May 5, 19	
COMPONENTS AT KE	NNEDY SPACE CEN	TER	6. PERFORMING ORG	ANIZATION CODE
7. AUTHOR (S) R. K. Dumbauld and J. R. Bjo	rklund		B. PERFORMING ORGA	NIZATION REPORT #
9. PERFORMING ORGANIZATION NAME AND AL	DDRESS		10. WORK UNIT NO.	
GCA Technology Division				
GCA Corporation			11. CONTRACT OR GE	RANT NO.
Bedford Park, Massachusetts			NAS8-26673	
12. SPONSORING AGENCY NAME AND ADDRES	S		13. TYPE OF REPORT	& PERIOD COVERED
			CONTRACTOR	REPORT
National Aeronautics and Spa	ce Administrati	on	000.00000000000000000000000000000000000	
Washington, D. C. 20546			14. SPONSORING AG	ENCY CODE
		or was Mr. John W		
Environment Division, Aero-As	strodynamics La	ooratory, Marshal	ll Space Flight	Center.
16. ABSTPACT				
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19. SECURITY CLASSIF. (of this report)	20. SECURITY CLAS	SIF. (of this page)	21. NO. OF PAGES	22. PRICE
Unclassified	Unclassified	l	66	\$3.00
MSFC - Form 3292 (May 1969)				<u> </u>

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#### INTRODUCTION

Estimates have been made of hazard distances downwind from normal launches of Saturn V type aerospace vehicles at Cape Kennedy for selected rocket fuel components under three meteorological situations. The selected fuel components, the maximum allowable concentrations for a ten-minute exposure period  $(MAC_{10})$  considered in estimating the hazard distances and the molecular weights of the components are given in Table 1-1.

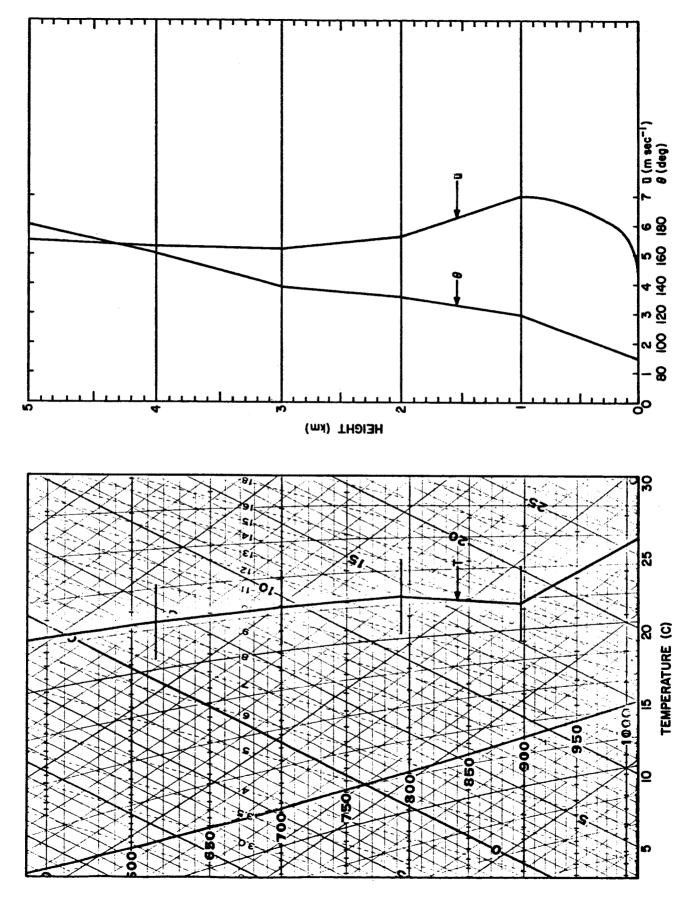
The three meteorological situations used for the hazard calculations are based on the mean monthly wind speed, wind direction, and temperature profiles for Kennedy Space Center (KSC) published by Smith and Vaughan (1961) and on the work of Record, et al. (1970). Preliminary concentration calculations showed that downwind hazard distances for normal launches of Saturn V type vehicles were primarily dependent on the depth of the surface mixing layer  $H_{\mathbf{m}}$ and the vertical distribution of material in the stabilized cloud of exhaust products. Study of the mean wind speed, wind direction, and temperature profiles for KSC showed that, for easterly wind flow required to transport the combustion product cloud inland, the average surface mixing depth was about 1000 meters. Two exceptions were noted. During the spring, there are a few occasions when the surface mixing depth is about 2000 meters; also, during the afternoon sea breeze in all seasons, the average surface mixing depth is about 300 meters. Figures 1-1, 1-2, and 1-3 show composite vertical profiles of air temperature, wind direction, and wind speed for easterly wind regimes in the fall and spring and for the afternoon sea breeze.

TABLE 1-1

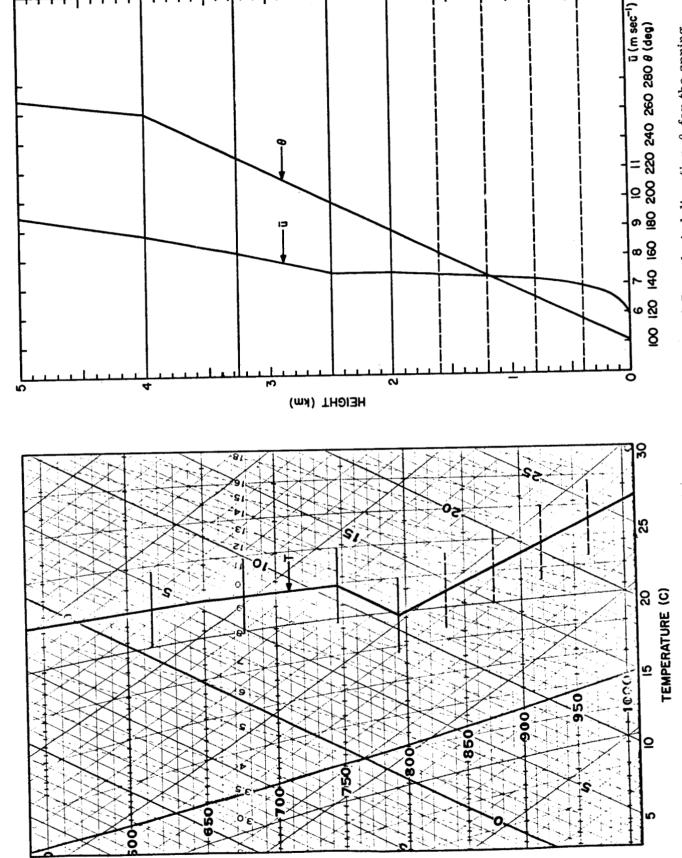
MOLECULAR WEIGHTS AND 10-MINUTE MAXIMUM
ALLOWABLE CONCENTRATIONS (MAC<sub>10</sub>) FOR
SELECTED ROCKET FUEL COMPONENTS

Fuel	Molecular	MAC <sub>10</sub> *
Component	Weight	(ppm)
со	28.01	1500
HC1	36.47	30
NO <sub>2</sub>	46.01	30
N <sub>2</sub> O <sub>4</sub>	92.02	30
N <sub>2</sub> H <sub>4</sub>	32.05	30
MMH	46.07	90
UDMH	60.00	100
$^{\mathrm{F}}_{2}$	38.00	15
нғ	20.01	20
Al <sub>2</sub> O <sub>3</sub>	101.94	50 (mg m <sup>-3</sup> )

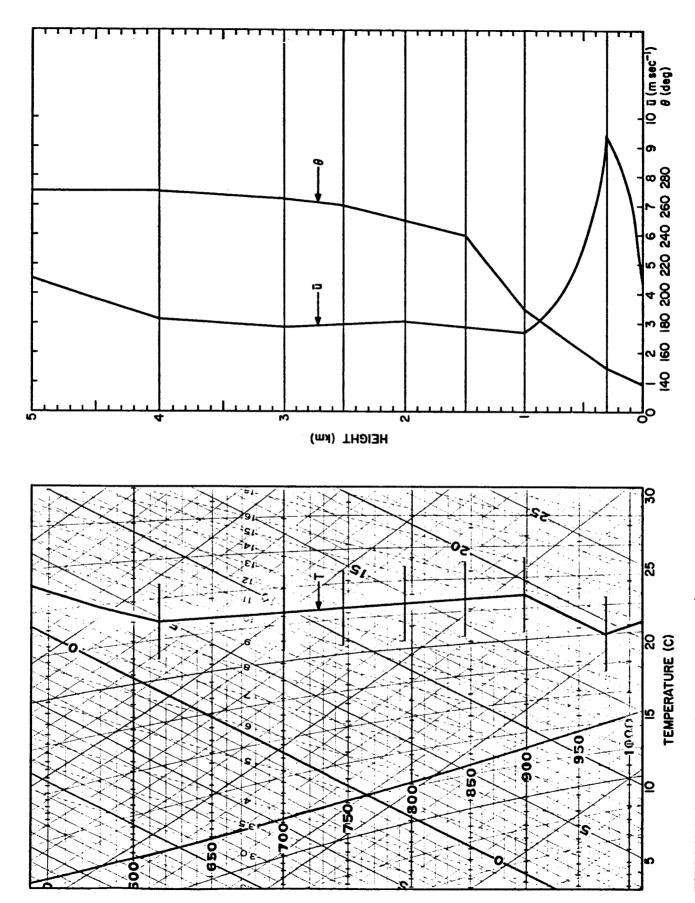
<sup>\*</sup>Supplied by personnel of Atmospheric Dynamics Branch, Aerospace Environment Division, Aero-Astrodynamics Laboratory, George C. Marshall Space Flight Center, Marshall Space Flight Center, Alabama



Vertical profiles of air temperature  $T_{2}$  mean wind direction  $\theta$ , and mean wind speed  $\bar{u}$  for the fall meteorological regime at KSC. Heavy horizontal lines indicate layer boundaries. FIGURE 1-1.



meteorological regime at KSC. Heavy solid and dashed horizontal lines indicate layer boundaries. Vertical profiles of air temperature T, mean wind speed  $\bar{\mathbf{u}}$ , and wind direction  $\theta$  for the spring FIGURE 1-2.



Vertical profiles of air temperature T, mean wind speed  $\vec{u}$ , and wind direction  $\theta$  for the afternoon sea-breeze regime at KSC. Heavy horizontal lines indicate layer boundaries. FIGURE 1-3.

#### CLOUD RISE CALCULATIONS

Estimates of maximum cloud rise for use in the hazard calculations were obtained from an expression due to Briggs (1969, p. 33; 1970):

$$z = \left[ \frac{3F_{m}}{\bar{u} \gamma^{2} s^{1/2}} \sin\left(s^{1/2} t\right) + \frac{3F}{\bar{u} \gamma^{2} s} \left(1 - \cos\left(s^{1/2} t\right)\right) \right]^{1/3}$$
 (2-1)

where

z = height of cloud at time t

$$F_{m} = w_{o}^{2} r_{o}^{2}$$

 $w_0 = initial vertical velocity (m sec^{-1})$ 

 $r_{o}$  = radius of area where vertical velocity equals  $w_{o}$  (m)

 $\bar{u}$  = mean wind speed (m sec<sup>-1</sup>)

 $\gamma$  = entrainment constant where cloud radius  $r\{z\} = \gamma z$ 

 $s = stability parameter = \frac{g}{T} \frac{\partial \Phi}{\partial z}$ 

 $g = gravitational acceleration = 9.8 m sec^{-2}$ 

T = ambient air temperature (OK)

 $\frac{\partial \Phi}{\partial z}$  = vertical potential temperature gradient (oK m<sup>-1</sup>)

$$F = \frac{g Q_H}{\pi c_p \rho T}$$

 $Q_{H}$  = heat emission due to efflux of hot gases (cal sec<sup>-1</sup>)

 $c_p$  = specific heat of air = 0.24 cal g<sup>-1</sup> o<sub>K</sub><sup>-1</sup>

 $\rho = \frac{\rho}{\rho} = \frac{3}{\rho}$ 

t = time (sec)

The above formula yields cloud rise rates that agree favorably with measured cloud rise rates from static tests of Saturn type engines published by Susko, Kaufman, and Hill (1968).

Input parameters used in the calculation of maximum cloud rise are shown in Table 2-1. The value of  $\mathbf{Q}_{\mathbf{H}}$  in the table was calculated from the expression

$$Q_H = c_{pm} T_m W$$

where

 $c_{pm}$  = specific heat of exhaust gas = 0.49 cal g<sup>-1</sup> o<sub>K</sub><sup>-1</sup>

 $T_{m} = \text{temperature of exhaust gas} = 2080^{\circ} \text{K}$ 

W = mass of exhaust gas burned per unit time =  $2.268 \times 10^6 \text{ g sec}^{-1} \text{ engine}^{-1}$ 

Thus, for the five F-1 engines as employed on the first stage of Saturn V, the value for  $Q_H$  equals  $1.155 \times 10^{10}$  cal sec<sup>-1</sup>. The values of  $c_{pm}$ ,  $T_{m}$ , W, and w were obtained from a report by Thayer, Chandler and Chu (1970). A value of 0.5 for the entrainment constant  $\gamma$  was chosen because Dumbauld (1971) found that this value provided the best fit of Equation (2-1) to the rate of cloud rise data from static engine tests mentioned above. Values for the meteorological parameters in Table 2-1 were obtained from Figures 1-1, 1-2, and 1-3.

The maximum cloud rise calculated from Equation (2-1) for the fall, spring, and sea-breeze meteorological regimes is, respectively, 2270, 2200, and 2260 meters.

It should be mentioned that, although the rate of cloud rise is highly dependent on the value of parameters in the first term of Equation (2-1), the

TABLE 2-1
INPUT PARAMETERS TO PLUME RISE FORMULA

	(a) Ger	neral	· · · · · · · · · · · · · · · · · · ·
$w_o^* = 488 \text{ m sec}^{-1}$ $r_o = 5 \text{ m}$ $\rho = 1190 \text{ g m}^{-3}$		$Q_{H}^{*} = 1.155 \text{ m}$ $\gamma = 0.5$ $c_{p} = 0.24 \text{ ca}$	c 10 <sup>10</sup> cal sec <sup>-1</sup>
	(b) Spe	cific	
	Meteorological Regime		
	Fall	Spring	Sea Breeze
T ( <sup>0</sup> K) ∂Φ/∂z ( <sup>0</sup> K m <sup>-1</sup> )	299 0.0044	300 0.0040	294 0.0064
$\bar{u} \text{ (m sec}^{-1})$	6.0	7.2	4.2
rom report by Thayer,	Chandler, and	Chu (1970)	

maximum cloud rise depends more on the values assigned to the parameters in the second term. The maximum cloud rise  $\ z_m$  can thus be estimated with sufficient accuracy from the expression

$$z_{\rm m} = \left[\frac{6F}{\bar{u} \gamma^2 s}\right]^{1/3} \tag{2-2}$$

# VERTICAL DISTRIBUTION OF ROCKET COMBUSTION PRODUCTS AND INITIAL CLOUD DIMENSIONS

The material in the cloud of combustion products from a normal launch is composed of two parts:

- The contribution from the first 30 seconds' operation of the rocket motors which is assumed to be entirely contained in the stabilized ground cloud
- The contribution from the operation of the rocket motors as the vehicle ascends from a height of 1236 meters (the height after 30 seconds) to 5 kilometers (the height after 56 seconds)

The geometry of the stabilized cloud of combustion products for the fall, spring, and sea-breeze meteorological regimes is shown in Figures 3-1, 3-2, and 3-3. The slanted solid lines between ground level and the height of maximum cloud rise show the diameter of the stabilized ground cloud calculated from the expression

$$r\{z\} = \gamma \left(z - z_0 + r_0/\gamma\right); 0 \le z \le z_m$$
 (3-1)

where

 $r_0$  = initial radius of the cloud at height z

 $z_{m}$  = height of maximum cloud rise

The entrainment parameter, as before, is set equal to 0.5. The initial radius  $r_0$  and height  $z_0$  were set equal to 454.3 meters. These values of  $r_0$  and  $z_0$  are consistent with cloud dimensions determined 30 seconds after ignition from time

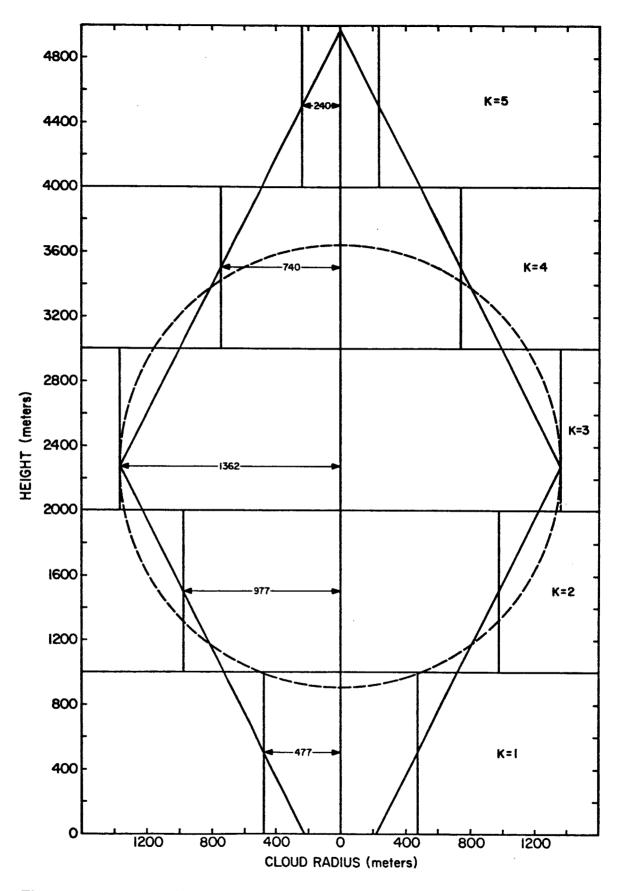


FIGURE 3-1. Geometry of the cloud of combustion products between ground level and 5 kilometers for the fall meteorological regime.

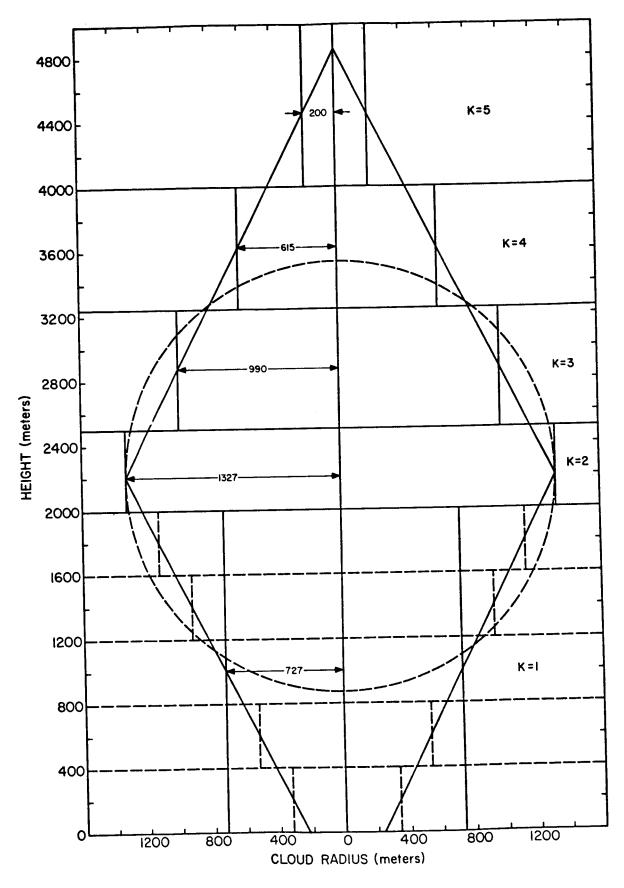


FIGURE 3-2. Geometry of the cloud of combustion products between ground level and 5 kilometers for the spring meteorological regime.

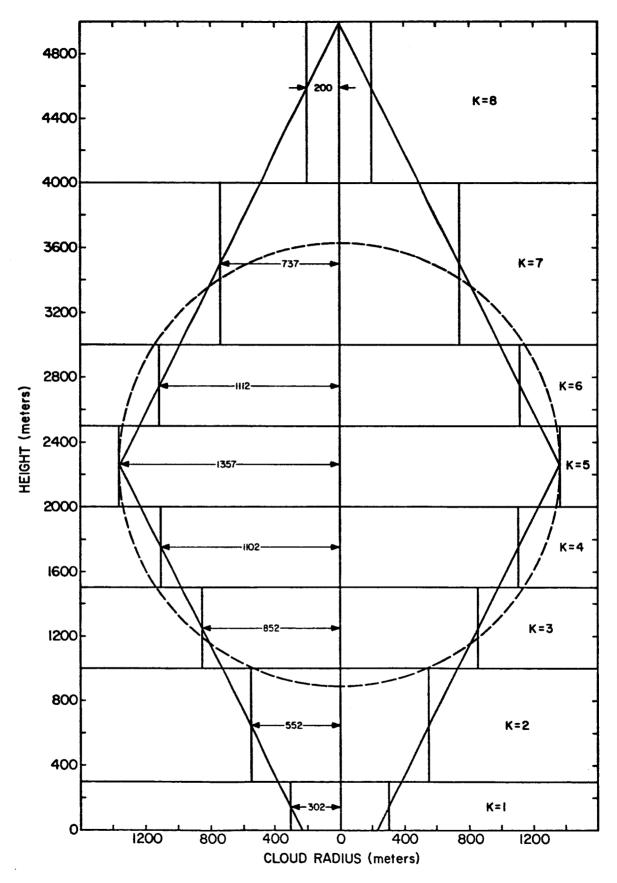


FIGURE 3-3. Geometry of the cloud of combustion products between ground level and 5 kilometers for the sea-breeze meteorological regime.

sequence photographs of the launch of Apollo 10. The slanted solid lines defining the cloud diameter above  $\mathbf{z}_{\mathbf{m}}$  are constructed from the expression

$$r\{z\} = z_m - \gamma \left(z + z_0 - \frac{r_0}{\gamma}\right) \ge 200; \quad z \ge z_m$$
 (3-2)

and are mirror images of the lines defining the cloud diameters below  $z_m$ . It should be noted that  $r\{z\}$  in this case is not permitted to become less than 200 meters.

The solid horizontal lines in Figures 3-1, 3-2, and 3-3 show the layer structure of the atmosphere based on the variations of meteorological profile quantities described in Section 1. The dashed horizontal lines shown in Figure 3-2 for the spring meteorological regime indicate an alternate division of the layer structure used to estimate ground-level concentrations.

The dashed circle in Figures 3-1, 3-2, and 3-3 shows the 2.15  $\sigma$  limits for a Gaussian distribution centered at  $z_m$ , where the standard deviation  $\sigma$  of the distribution is defined by

$$\sigma = \frac{\mathbf{r} \left\{ \mathbf{z} = \mathbf{z}_{\mathbf{m}} \right\}}{2.15} \tag{3-3}$$

The fraction of material  $F_g$  in each layer from the initial 30 seconds operation of the rocket motors is given by the expression

$$F_{g} = \left\{ P\{z_{TK}\} \left[ \frac{z_{TK} - z_{max}}{\sigma} \right] - P\{z_{BK}\} \left[ \frac{z_{BK} - z_{max}}{\sigma} \right] \right\}$$
(3-4)

where

 $F_g$  = fraction of material from initial 30 seconds' emission found in the layer

 $P\{\,^{Z}_{TK}\} = \text{integral of the Gaussian (normal) probability function between minus infinity and the top of the $K^{th}$ layer $^{Z}_{TK}$ }$ 

 $P\{^{z}_{BK}\} = \underset{\text{between minus infinity and the bottom of the $K^{th}$ layer}{\text{BK}}$ 

Material is also added in the lowest 5 kilometers of the atmosphere by vehicle emission subsequent to the initial 30 seconds. Figure E-7 of the NASA report by Dumbauld, et al. (1970) describes the altitude of the Saturn vehicle as a function of time after ignition. The time required for the Saturn vehicle to reach a given altitude  $\leq$  5 kilometers (t  $\leq$  56 seconds) is given by the expression

$$t_{v} = 1.25 (z)^{0.44636}$$
;  $z \le 5000 m$  (3-5)

where

t = time after ignition (sec)

z = Saturn altitude (m)

The total fraction  $\mathbf{F}_{T}$  of material in any layer resulting from Saturn emissions over the first 56 seconds of flight, during which the vehicle reaches an altitude of 5 kilometers, is given by the expressions

$$F_{TK} = \left\langle \frac{30}{56} \text{ F} + \left( \frac{1.25 \text{ z}_{TK}^{0.44636} - 30}{56} \right) \right. ; z_{TK} > 1,236 \text{m}, z_{BK} < 1,236 \text{m} \right\rangle (3-6)$$

$$\left\langle \frac{30}{56} \text{ F} + \frac{1.25}{56} \left( z_{TK}^{0.44636} - z_{BK}^{0.44636} \right) ; z_{TK} > 1,236 \text{m}, z_{BK} > 1,236 \text{m} \right\rangle (3-6)$$

Table 3-1 gives values of  $F_{TK}$  calculated from Equation (3-6) for the three meteorological regimes. The two columns under the spring meteorological regime represent the distribution of material based on the meteorological layer structure (1) and the additional layer structure for better definition of ground-level concentration (2).

The initial lateral  $\sigma_{yo}\{K\}$  and alongwind  $\sigma_{xo}\{K\}$  cloud dimensions for the  $K^{th}$  layer are determined from the average radius of the ground cloud in the layer. The average cloud diameters are shown in Figures 3-1, 3-2, and 3-3 as solid vertical lines and the corresponding cloud radii are given by the numerical values printed in the figures. The expressions used to calculate the cloud dimensions are

$$\sigma_{yo}\{K\} = \sigma_{xo}\{K\} = \left\{H - \gamma \left[\left(\frac{z_{TK} + z_{BK}}{2}\right) - \frac{r_o}{\gamma} + z_o\right]\right\} / 2.15$$
 (3-7)

when the K<sup>th</sup> layer is above the layer containing z<sub>m</sub>;

$$\sigma_{yo}\{K\} = \sigma_{xo}\{K\} = \left\{\gamma \left[z_m + \frac{r_o}{\gamma} - z_o\right]\right\} / 2.15$$
 (3-8)

when the K<sup>th</sup> layer contains z<sub>m</sub>; and

$$\sigma_{yo}\{K\} = \sigma_{xo}\{K\} = \left\{ \gamma \left[ \left( \frac{z_{TK} - z_{BK}}{2} \right) + \frac{r_o}{\gamma} - z_o \right] \right\} / 2.15$$
 (3-9)

when the  $K^{th}$  layer is below the layer containing  $z_m$ . The minimum initial cloud dimensions permitted in any layer are

$$\sigma_{yo}\{K\} = \sigma_{xo}\{K\} = \frac{200}{2.15} = 93 \text{ meters}$$
 (3-10)

The initial cloud dimensions calculated for the three meteorological regimes calculated from Equations (3-7) through (3-10) are given in Table 3-2.

TABLE 3-1  $\begin{tabular}{ll} TOTAL FRACTION OF MATERIAL ($F_{TK}$) IN THE $K^{th}$ LAYER FROM SATURN EXHAUST EMISSIONS* \\ \end{tabular}$ 

-		Meteorolo	gical Regime	
Layer (K)	Fall		ring	Sea Breeze
(/		(1)	(2)	
1	0.012	0.328	0.0009	0.0005
2	0.296	0.238	0.005	0.012
3	0.421	0.235	0.022	0.097
4	0.174	0.103	0.126	0.201
5	0.096	0.096	0.174	0.234
6			0.238	0.186
7			0.235	0.172
8			0.103	0.097
9			0.096	

<sup>\*</sup>Fractions may not add to unity because of round-off errors.

TABLE 3-2 INITIAL CLOUD DIMENSIONS IN THE K<sup>th</sup> LAYER  $\left(\sigma_{xo}\{K\} = \sigma_{yo}\{K\} \text{ in meters}\right)$ 

_		Meteorologica	l Regime	
Layer	Fall	Fall Spring		Sea Breeze
(K)	Fall	(1)	(2)	
1	222	338	152	140
2	454	617	245	257
3	632	461	338	396
4	344	286	431	513
5	112	93	524	631
6			617	517
7			461	343
8			286	93
9			93	

#### CONCENTRATION MODELS

With exception of the dispersion model for calculating time-averaged concentrations, the models described in this section are documented in detail in a report prepared for NASA-Huntsville (Dumbauld, et al., 1970). A description of the time-averaged concentration model is included in a revised version of the above-mentioned report to be submitted to NASA in May 1971. For these reasons, only a brief description of the models is given in this section.

Peak concentrations in the  $K^{th}$  layer along the axis of cloud travel were calculated for the three meteorological regimes described above from the expression

$$\chi_{\mathbf{p}} = \frac{Q_{\mathbf{K}}}{2\pi \sigma_{\mathbf{y}\mathbf{K}} \sigma_{\mathbf{x}\mathbf{K}}}$$
 (4-1)

where

 $\mathbf{Q}_{K}^{}$  = source strength in units of mass per unit depth of the  $K^{th}$  layer

 $\sigma_{yK} \ = \ \frac{\text{standard deviation of the crosswind concentration}}{\text{distribution in the } K^{th} \ \text{layer}}$ 

 $\sigma_{xK}$  = standard deviation of the alongwind concentration distribution in the  $K^{th}$  layer

Equation (4-1) above and the subset of equations defining  $\sigma_{yK}$  and  $\sigma_{xK}$  are given on pages 14 through 20 of the report by Dumbauld, et al. (1970). The use of Equation (4-1) implies that material originating in the  $K^{th}$  layer is constrained from diffusing vertically beyond the vertical boundaries of that layer.

Estimates were also made of ground-level peak concentration for the spring meteorological regime where the lowest 2 kilometers of the atmosphere is

further subdivided into layers based on the vertical distribution of material (see columns labeled (2) in Tables 3-1 and 3-2). In this case, material contained in the layers below 2 kilometers indicated by horizontal dashed lines in Figure 3-2 must be permitted to diffuse vertically into the surface mixing layer because there is no meteorological restriction to mixing in the lowest 2 kilometers. The layer transition model described as Model 5 by Dumbauld, et al. (1970, pp. 31-33) proves to be a useful mathematical artifice for permitting the requisite vertical mixing while identifying the initial vertical structure of material. In the model, material in the original K layers is permitted to diffuse into a new layer (in this case the lowest 2 kilometers) at a predetermined time t\*. For these calculations t\* was set equal to 1 second. The hazard estimates using the layer transition model are compared to those obtained using Equation (4-1) for the spring meteorological regime in Section 6.

Time-average concentrations were calculated from the expression

$$\chi_{\text{ave}} = \frac{\chi_{\text{p}} \sqrt{2\pi} \sigma_{\text{xK}}}{\bar{u}_{\text{K}} T_{\text{A}}} \left\{ \text{erf} \left( \frac{\bar{u}_{\text{K}} T_{\text{A}}}{2\sqrt{2} \sigma_{\text{xK}}} \right) \right\}$$
 (4-2)

where

 $\bar{u}_{K}^{}$  = mean transport speed in the  $K^{th}$  layer

T<sub>A</sub> = time over which concentration is averaged in seconds

#### MODEL INPUTS

Initial cloud dimensions used in the hazard calculations are given in Table 3-2 and the vertical distributions of rocket exhaust products emitted during the first 56 seconds of a normal Saturn vehicle launch are given in Table 3-1. To simplify the computer calculations, the source strength in each layer was normalized. The results of the computer calculations must be multiplied by a simple scaling factor to obtain the concentration for a specific fuel component. Normalized source strengths shown in Table 5-1 were computed from the expression

$$Q_{K} = \frac{(F_{TK})^{(T)} 22.4 \times 10^{3}}{(z_{TK} - z_{BK})^{273.2}}$$
 (5-1)

where  $F_{TK}$  is the fractional percent in the  $K^{th}$  layer from Table 3-1, and  $\left(z_{TK}-z_{BK}\right)$  is the depth of the layer. The value of temperature T in the layer was set equal to 294 degrees for the fall and sea-breeze meteorological regimes and to 299 degrees for the spring meteorological regime.

Additional meteorological inputs are given in Table 5-2. Values of the mean wind speed  $\bar{u}_{BK}$  and wind direction  $\theta_{BK}$  at the base of the  $K^{th}$  layer were determined from the vertical profiles shown in Figures 1-1 through 1-3. Values of the standard deviation of the wind azimuth angle at the reference height  $z_R$  = 18 meters, for a 10-minute sampling period ( $\tau_{oK}$  = 600 seconds), were obtained from the expression

$$\sigma_{ABK} \left\{ \tau_{oK} = 600 \text{ sec} ; K = 1 \right\} = \frac{R_d}{6}$$
 (5-2)

where  $R_d$  is the wind direction range at the reference height  $z_R$  from Figure 2-11 of the report by Record, et al. (1970). The quantity  $\sigma_A$  was assumed to decrease

	Meteorologic	cal Regime		
Fall			Sea Breeze	
	(1)	(2)		
0.2905	4.019	0.0582	0.0402	
7.130	11.634	0.3245	0.4064	
10.160	7.684	1.341	4.687	
41.970	3.364	7.708	9.692	
23.350	2.354	10.660	11.301	
		11.634	8.973	
		7.684	4.147	
		3.364	2.331	
·		2.354		
	0.2905 7.130 10.160 41.970	Fall (1)  0.2905 4.019  7.130 11.634  10.160 7.684  41.970 3.364	(1)       (2)         0.2905       4.019       0.0582         7.130       11.634       0.3245         10.160       7.684       1.341         41.970       3.364       7.708         23.350       2.354       10.660         11.634       7.684         3.364       3.364	

ADDITIONAL METEOROLOGICAL INPUTS TO THE HAZARD CALCULATIONS TABLE 5-2

Rogima	Daromotor					Layer	er (K)				
omigon.	raiametei	1	2	3	4	5	9	7	∞	6	10
Fall	$ m u_{BK}~(m~sec^{-1})$	4.7	7.0	5.6	5.2	5.3	5.5				
	$^{ heta}_{ m BK}$ (deg)	06	119	131	138	161	180				
	$\sigma_{ extbf{ABK}}^{ extbf{T}_{ ext{OK}}}$ (deg)	12	œ	<del></del> 4	∺	₹-4	<del></del>				
Spring	$ar{u}_{ m BK}$ (m sec $^{-1}$ )	9	6.9	7.2	7.3	7.4	7.5	7.5	8.2	8.8	9.5
	$^{ heta}_{ m BK}$ (deg)	100	116	133	148	164	180	200	230	260	270
	$\sigma_{ extbf{ABK}}^{ extbf{T}_{ ext{OK}}}$ (deg)	7	6.04	5.85	5.74	5.66	5.60		H	H	₩
Sea Breeze	uBK (m sec <sup>-1</sup> )	4.5	9.2	2.7	2.9	3.1	3.0	2.9	3.2	4.5	
	$^{ heta}_{ m BK}$ (deg)	140	150	190	240	250	260	265	270	270	
	$\sigma_{ extbf{ABK}}^{ ext{ToK}}$ (deg)	12	5.7	1	<b>-</b>	Н	<b>H</b>	H	-	H	

with height according to the expression

$$\sigma_{A}\{z, K=1\} = \sigma_{ABK}\{z=z_{R}, K=1\} \left(\frac{z}{z_{R}}\right)^{-p}$$
 (5-3)

as suggested by Record, et al. (1970, p. 48).

The power-law exponent p in Equation (5-3) is given by the expression

$$p = \log \left( \frac{\bar{u}_{TK} \{K=1\}}{\bar{u}_{BK} \{z_{R}, K=1\}} \right) / \log \left( \frac{z_{TK} \{K=1\}}{z_{R}} \right)$$

Note that a reference height of  $z_R$  = 18 meters was used for all meteorological parameters in the surface mixing layer. In the next higher layer (K = 2), the value of  $\sigma_A$  was linearly decreased from the value at the top of the surface layer to a value of 1.0 degrees at the top of the K = 2 layer. In all higher layers (K > 2),  $\sigma_A$  was held constant at 1.0 degrees.

#### RESULTS OF THE HAZARD CALCULATIONS

Figures 6-1, 6-2, and 6-3 show normalized ground-level peak and tenminute average concentrations for the fall, spring, and sea-breeze meteorological regimes obtained from Equations (4-1) and (4-2) above. In each case, the normalized source strengths given in Table 5-1 were used in the calculations. The layer structure for these calculations is shown by the solid horizontal lines in Figures 3-1, 3-2, and 3-3. The normalized concentrations are given in units of parts per million mole per gram of material released in the first 56 seconds after ignition. To obtain the concentration in parts per million (ppm) for the fuel components listed in Table 1-1, the concentrations in the figures must be multiplied by the total amount of material in grams released in the first 56 seconds following ignition and divided by the molecular weight of the material in grams. For example, suppose that  $10^8$  grams of hydrogen fluoride (HF) were actually released in the lowest 5 kilometers in the fall season. The ten-minute average concentration at 10,000 meters from the source is found from Figure 6-1 by entering the graph at 10<sup>4</sup> meters and obtaining a normalized concentration of 1.9 x  $10^{-8}$  ppm mole  $g^{-1}$ ; multiplying by the source strength in grams; and dividing by the molecular weight of HF (20.008). Thus,

$$\chi_{10 \text{ min}}^{(HF)} = \frac{1.9 \times 10^{-8} (10^{8})}{20.008} = 9.5 \times 10^{-2} \text{ ppm}$$

Figure 6-4 shows normalized ground-level peak and ten-minute average concentrations for the spring meteorological regime, calculated by means of the layer transition model discussed in Section 4 above and the layer structure in the lowest 2 kilometers as indicated by the dashed horizontal lines in Figure 3-2. In contrast to the normalized concentration shown in Figure 6-2 for the spring meteorological regime, the ground-level concentrations shown in Figure 6-4 are an order of

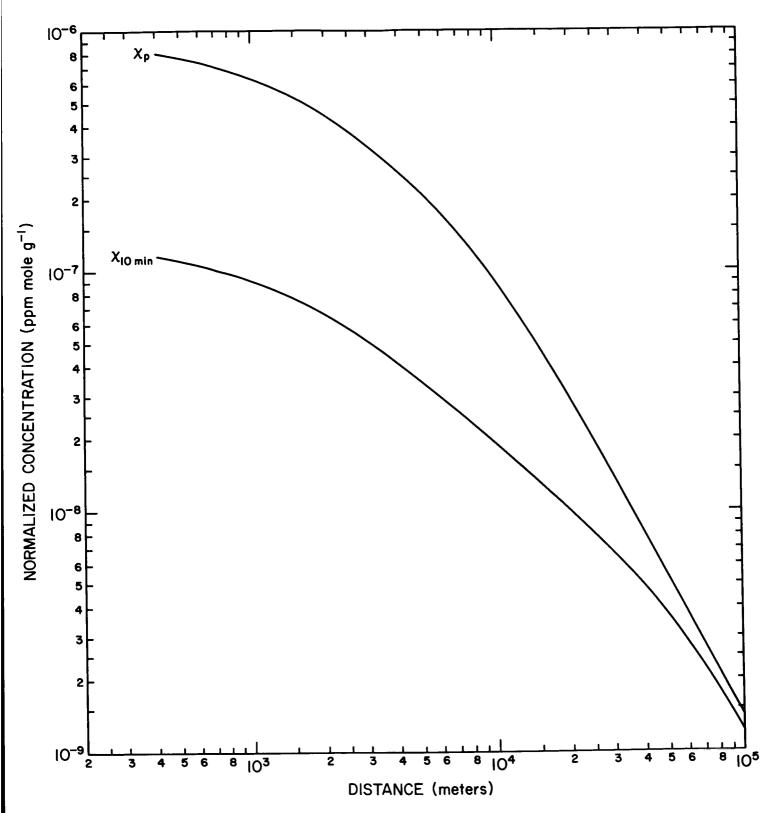


FIGURE 6-1. Normalized ground-level peak ( $\chi_p$ ) and ten-minute average ( $\chi_{10~min}$ ) concentrations downwind from a normal Saturn V launch for the fall-season meteorological regime. Source strength is one gram of material released in first 5 kilometers.

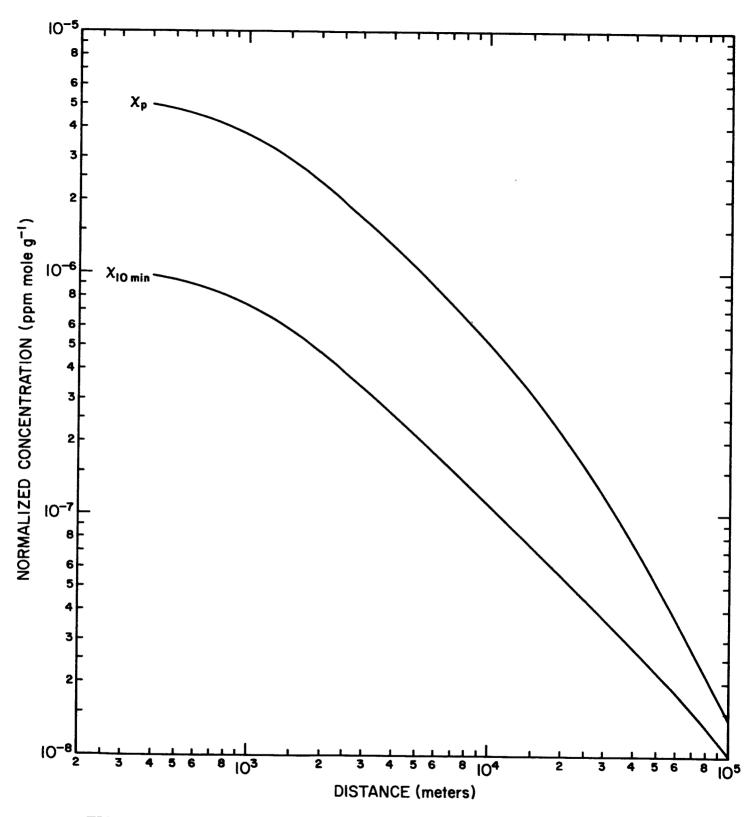


FIGURE 6-2. Normalized ground-level peak  $(\chi_p)$  and ten-minute average  $(\chi_{10 \text{ min}})$  concentrations downwind from a normal Saturn V launch for the spring meteorological regime. Source strength is one gram of material released in first 5 kilometers.

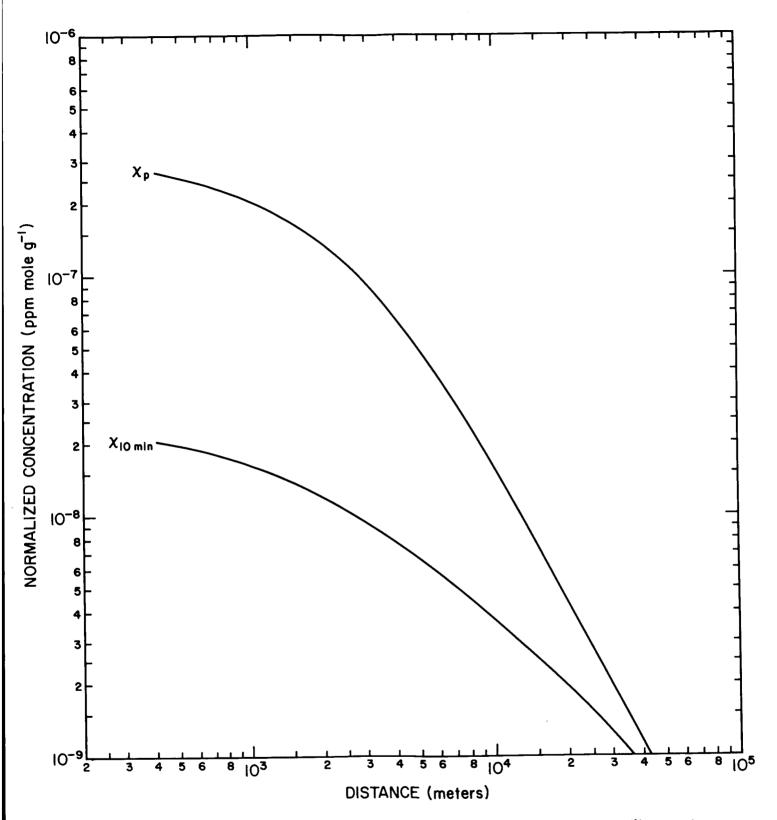


FIGURE 6-3. Normalized ground-level peak  $(\chi_p)$  and ten-minute average  $(\chi_{10 \text{ min}})$  concentrations downwind from a normal Saturn V launch for the seabreeze meteorological regime. Source strength is one gram of material released in first 5 kilometers.

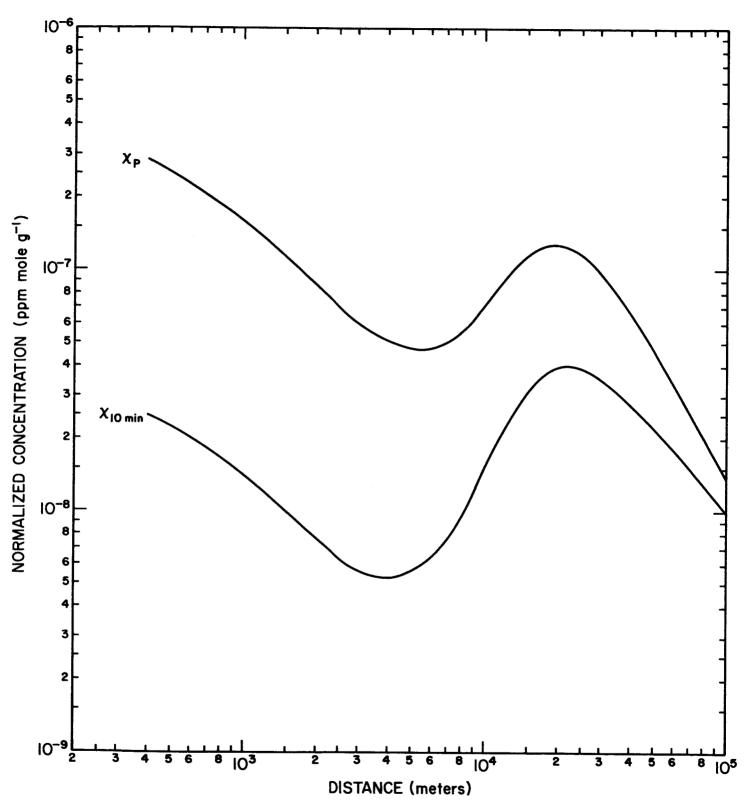


FIGURE 6-4. Normalized ground-level peak  $(\chi_p)$  and ten-minute average  $(\chi_{10-min})$  concentrations downwind from a normal Saturn V launch for the spring meteorological regime. Calculations were made using the layer transition model. Source strength is one gram of material released in first 5 kilometers.

magnitude lower in the first 4 kilometers downwind from the source. Beyond 4 kilometers, the concentrations shown in Figure 6-4 increase to a distance of about 25 kilometers from the source at which point they become nearly equivalent to the concentrations shown in Figure 6-2. The lower concentrations near the source predicted by the layer transition model are caused by the high initial concentration of material at the top of the surface mixing layer. In using the layer transition model, the material near the top of the layer is brought to the surface through turbulent mixing. As shown in Figure 6-4, complete vertical mixing has occurred in the layer transition case when the cloud reaches a downwind distance of 30 kilometers. In calculating the normalized ground-level concentrations shown in Figure 6-2, it has been assumed that the material is completely mixed in the surface layer (K = 1) prior to downwind travel from the source.

Although the concentration estimates shown in Figure 6-4 for the spring meteorological regime are likely more realistic than those shown in Figure 6-2, an arbitrary decision was made to use the estimates shown in Figure 6-2 in developing the remaining hazard estimates described in this section. This decision was made primarily because the use of the estimates from Figure 6-2 leads to more conservative hazard estimates.

Figures 6-5, 6-6, and 6-7 also show plots of ground-level peak concentration and 10-minute average concentrations for the three meteorological regimes. These graphs are similar to those in Figures 6-1, 6-2, and 6-3, with the exception that the source strength used to normalize the figures is one pound of material released in the surface mixing layer. Concentrations in parts per million for specific materials can be obtained from the normalized values through multiplication by the source strength in the surface layer in pounds, and dividing by the appropriate molecular weight in grams. The surface mixing layer depths used in the calculations for the fall, spring, and sea-breeze regimes are: 1000, 2000, and 300 meters. Estimates of concentrations for other mixing layer depths can be

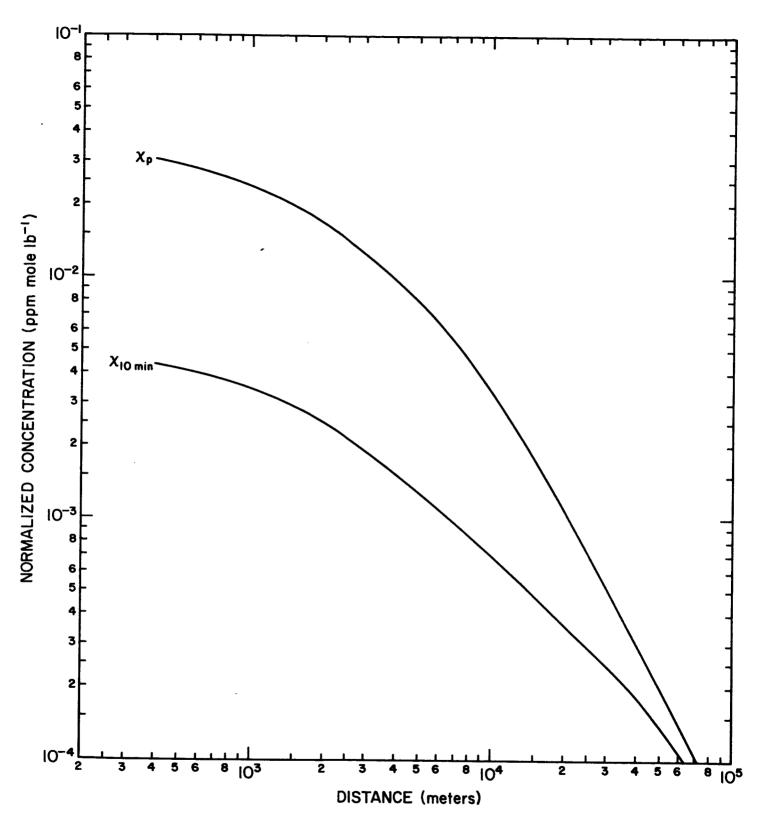


FIGURE 6-5. Normalized ground-level peak  $(\chi_p)$  and ten-minute average  $(\chi_{10~min})$  concentrations downwind from a normal Saturn V launch for fall season meteorological regime. Source strength is one pound of material released in surface mixing layer.

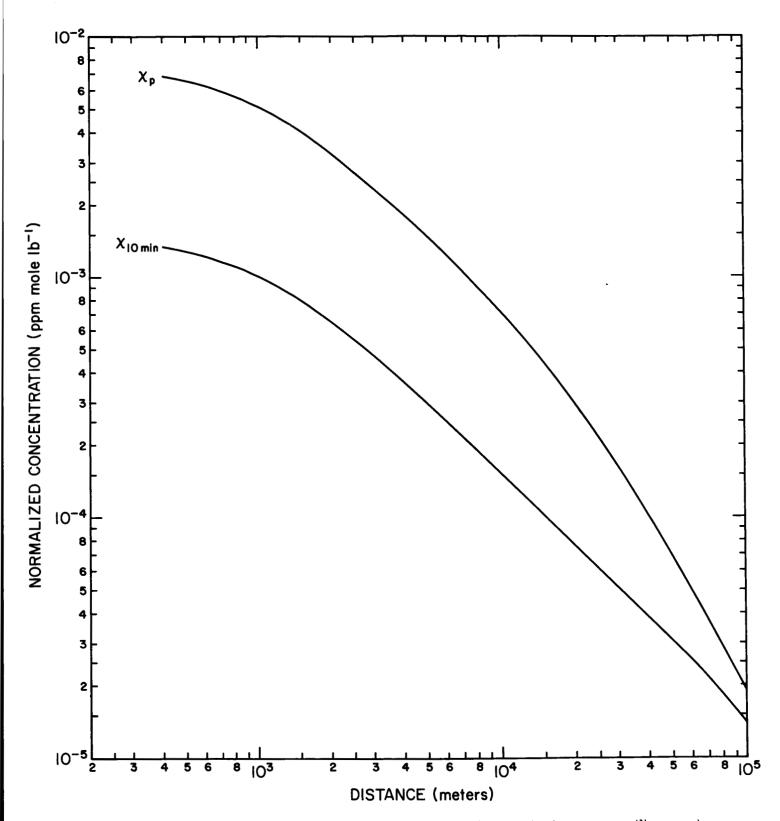


FIGURE 6-6. Normalized ground-level peak ( $\chi_p$ ) and ten-minute average ( $\chi_{10}$  min) concentrations downwind from a normal Saturn V launch for spring season meteorological regime. Source strength is one pound of material released in surface mixing layer.

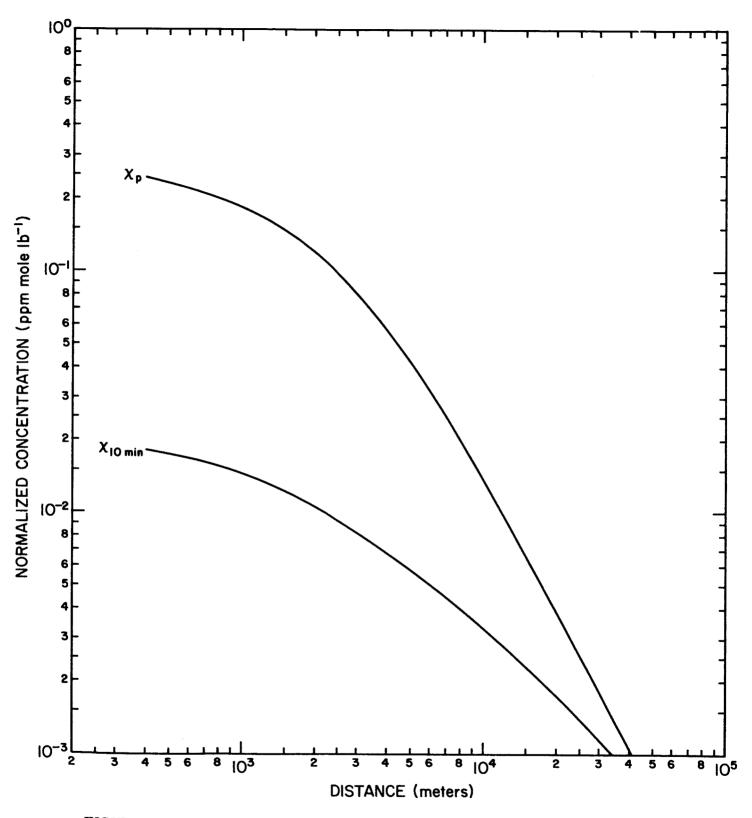


FIGURE 6-7. Normalized ground-level peak  $(\chi_p)$  and ten-minute average  $(\chi_{10~min})$  concentrations downwind from a normal Saturn V launch for sea-breeze meteorological regime. Source strength is one pound of material released in surface mixing layer.

obtained by multiplying the concentration estimates in Figures 6-5, 6-6, and 6-7. by the ratio of the mixing layer depths given above to the new mixing layer depths.

The hazard distance curves in Figures 6-8 and 6-9 were obtained from Figures 6-5, 6-6, and 6-7. Figure 6-8 shows the source strength in pounds required in the surface layer to produce a ground-level peak CO concentration of 1500 parts per million (MAC $_{10}$  from Table 1-1) at various downwind distances. For example, a source strength of 2.3 x  $_{10}$  pounds of CO in the surface mixing layer produces a peak concentration of 1500 parts per million at a distance of 1 kilometer from the source in the sea-breeze meteorological regime. Figure 6-9 shows the hazard distances and surface-layer source strengths for a ten-minute average CO concentration of 1500 parts per million.

Figures 6-10 through 6-27 show the weight in pounds of each specific material listed in Table 1-1 required to produce the  $MAC_{10}$  given in the table at distances greater than 400 meters from the source.

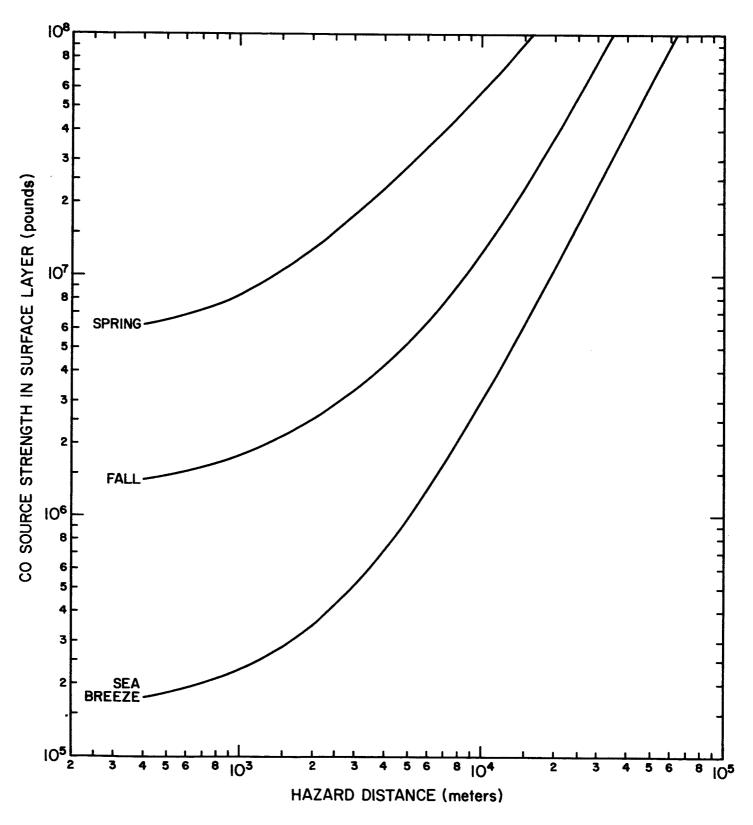


FIGURE 6-8. Hazard distances for a peak CO concentration of 1500 ppm for various surface layer source strengths.

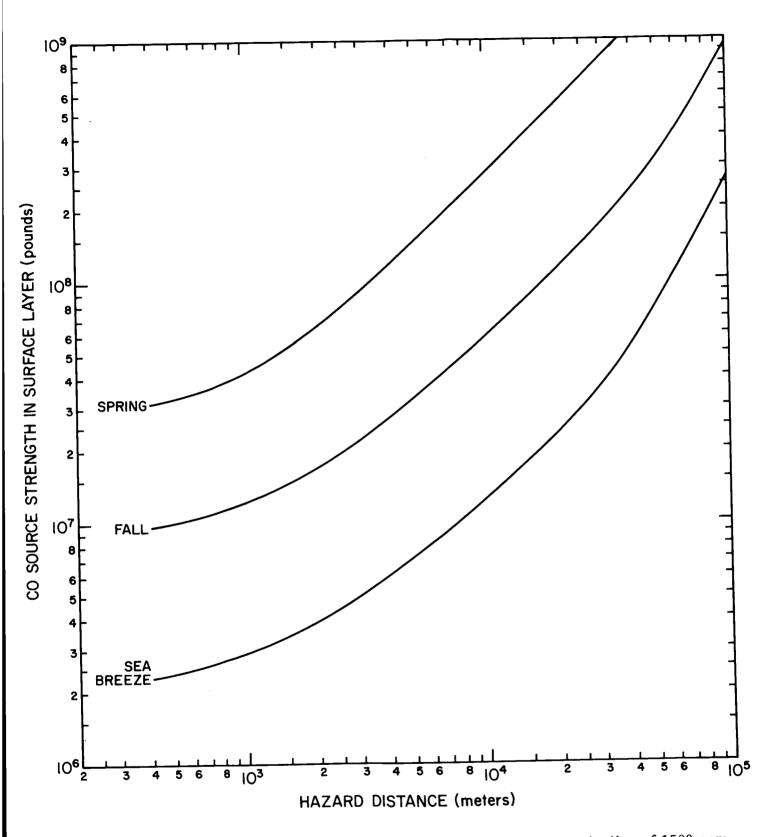


FIGURE 6-9. Hazard distances for a 10-minute average CO concentration of 1500 ppm for various surface layer source strengths.

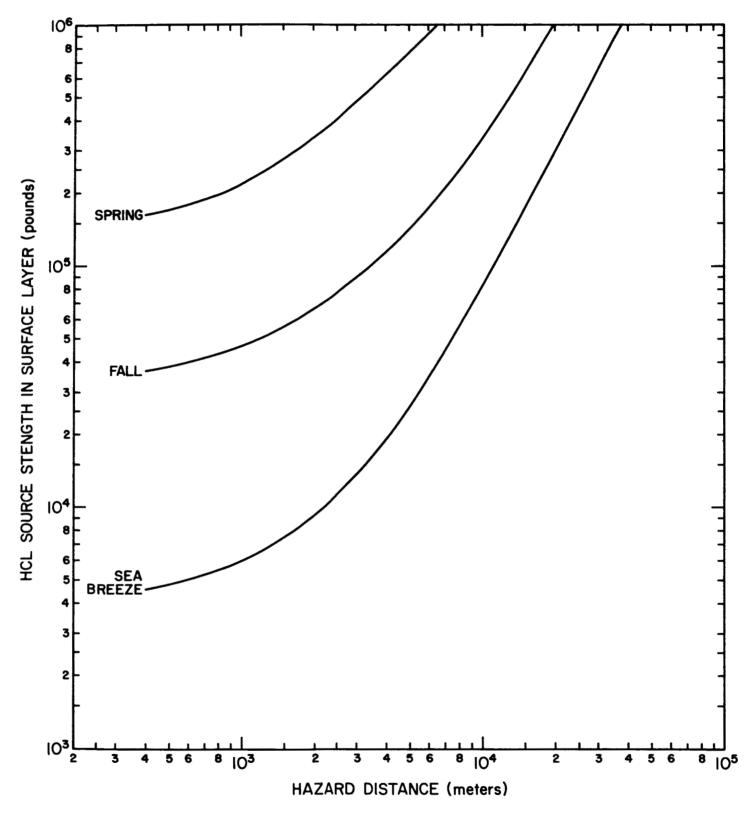


FIGURE 6-10. Hazard distances for a peak HCl concentration of 30 ppm for various surface-layer source strengths.

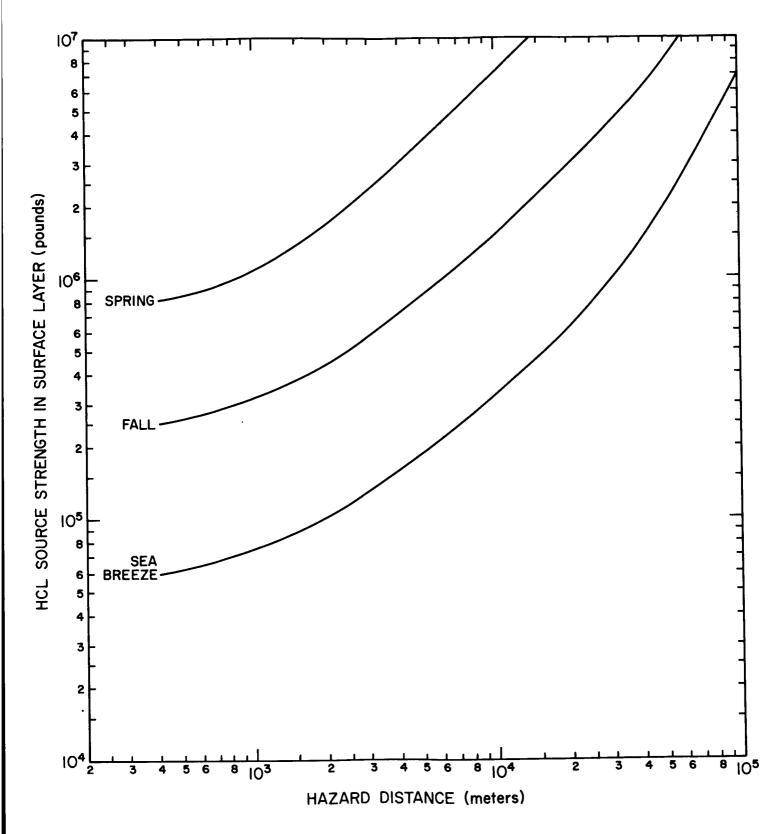


FIGURE 6-11. Hazard distances for a 10-minute average HCl concentration of 30 ppm for various surface-layer source strengths.

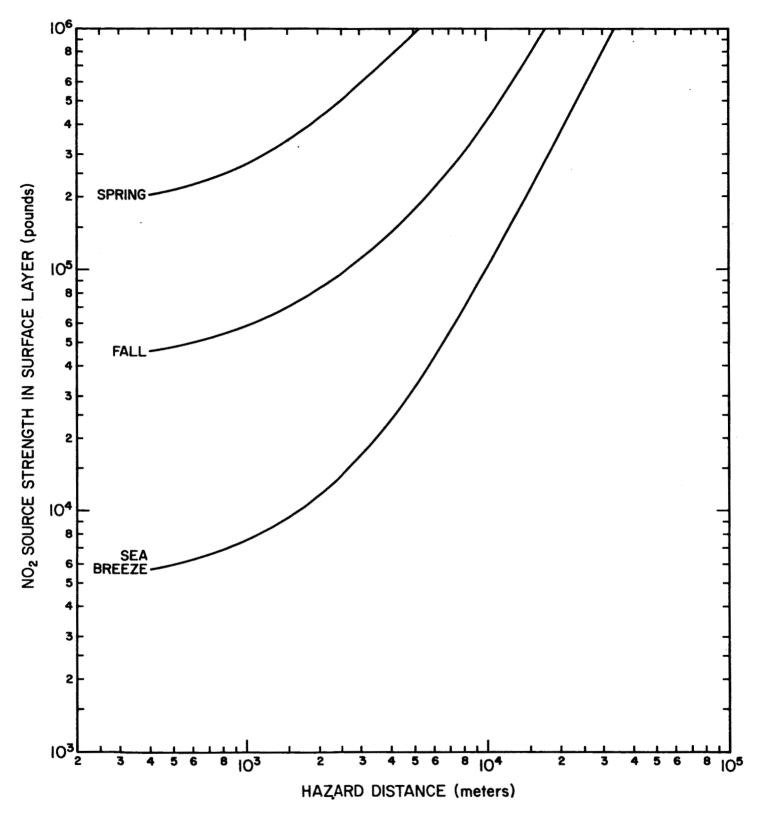


FIGURE 6-12. Hazard distances for a peak  $\rm NO_2$  concentration of 30 ppm for various surface-layer source strengths.

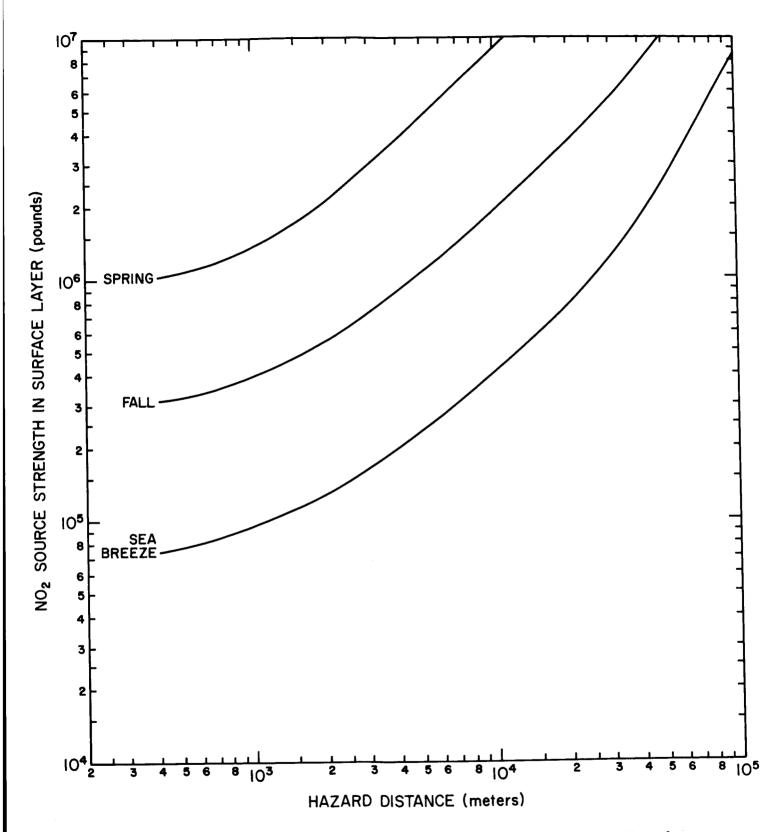


FIGURE 6-13. Hazard distances for a 10-minute average  $\rm NO_2$  concentration of 30 ppm for various surface-layer source strengths.

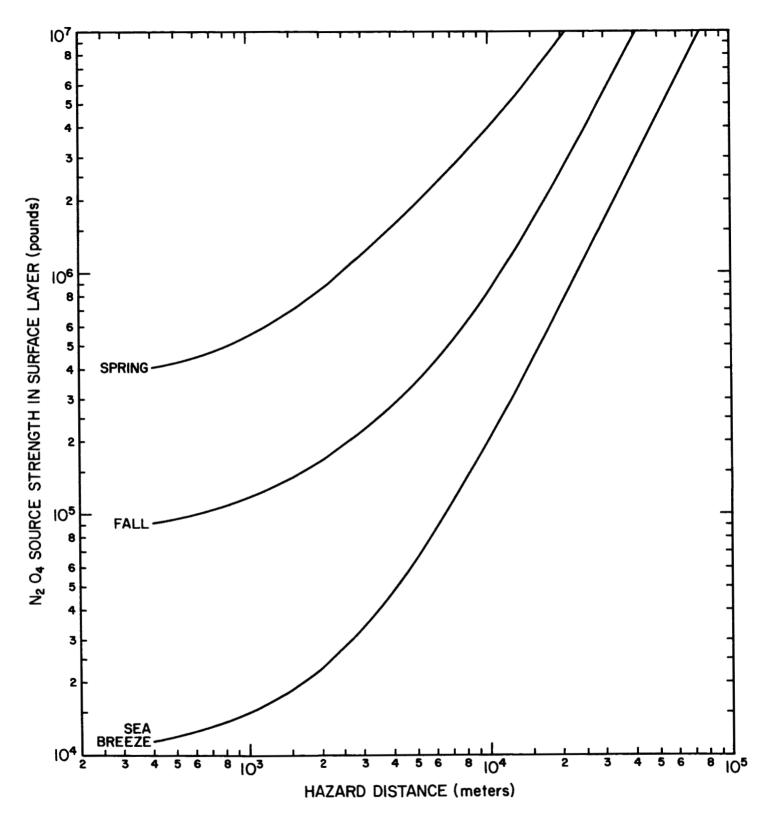


FIGURE 6-14. Hazard distances for a peak  $\rm N_2O_4$  concentration of 30 ppm for various surface-layer source strengths.

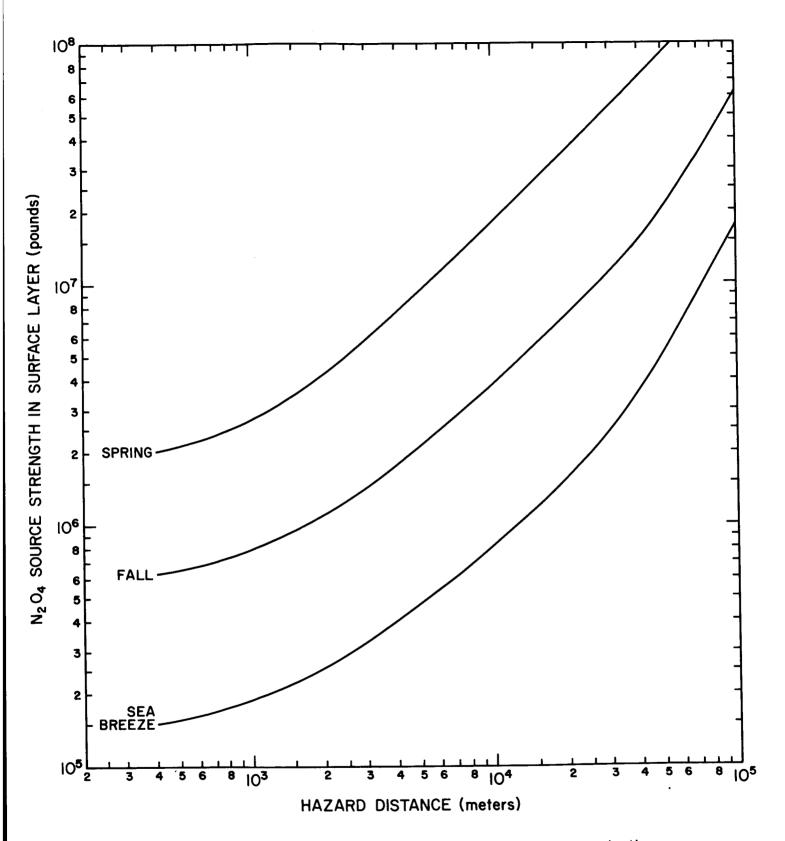


FIGURE 6-15. Hazard distances for a 10-minute average  $\rm N_2O_4$  concentration of 30 ppm for various surface-layer source strengths.

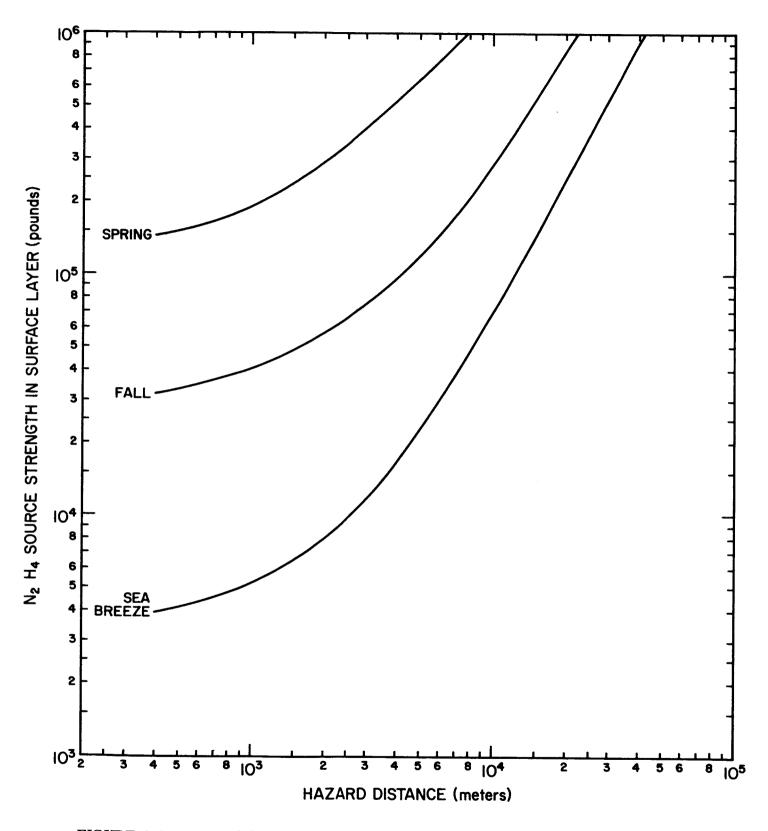


FIGURE 6-16. Hazard distances for a peak  $\rm N_2H_4$  concentration of 30 ppm for various surface-layer source strengths.

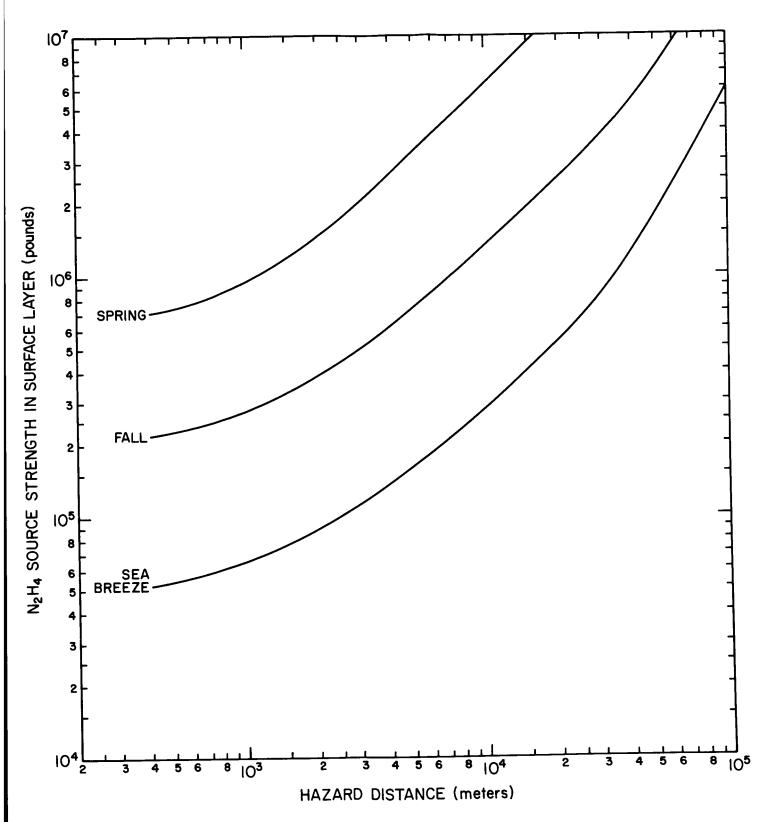


FIGURE 6-17. Hazard distances for a 10-minute average  $\rm N_2H_4$  concentration of 30 ppm for various surface-layer source strengths.

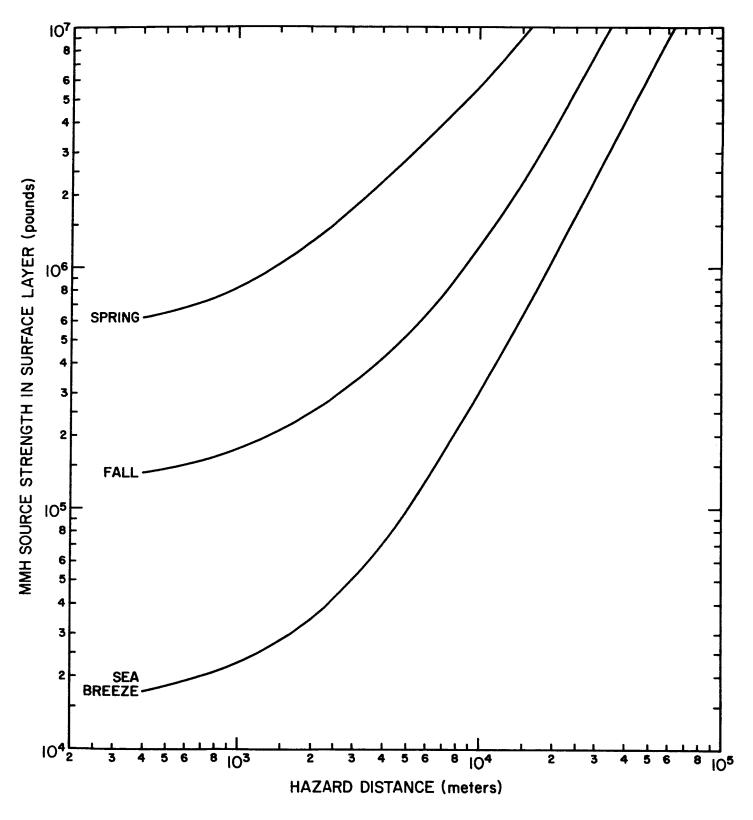


FIGURE 6-18. Hazard distances for a peak MMH concentration of 90 ppm for various surface-layer source strengths.

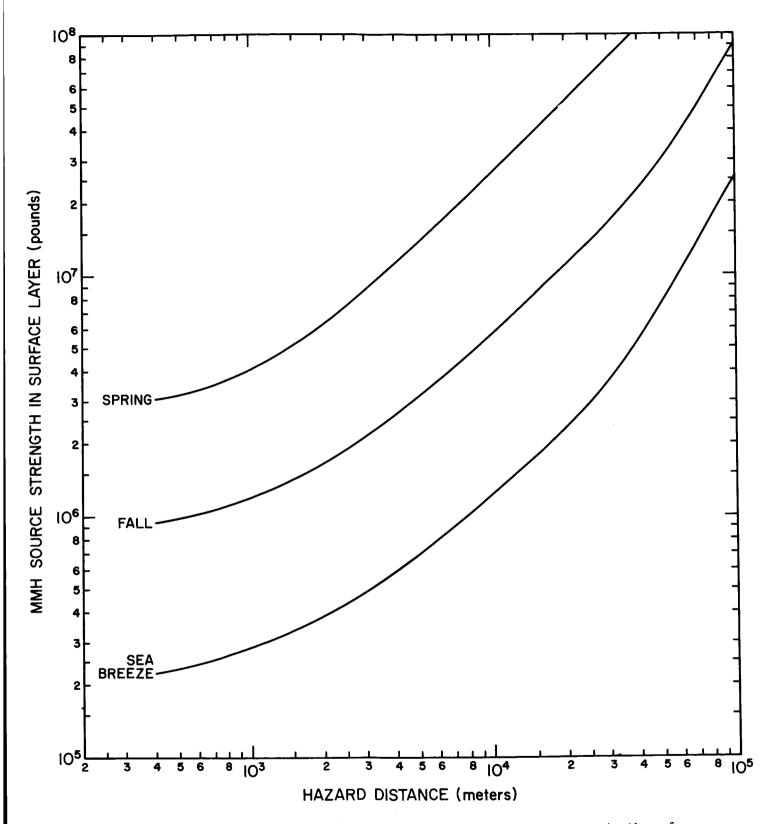


FIGURE 6-19. Hazard distances for a 10-minute average MMH concentration of 90 ppm for various surface-layer source strengths.

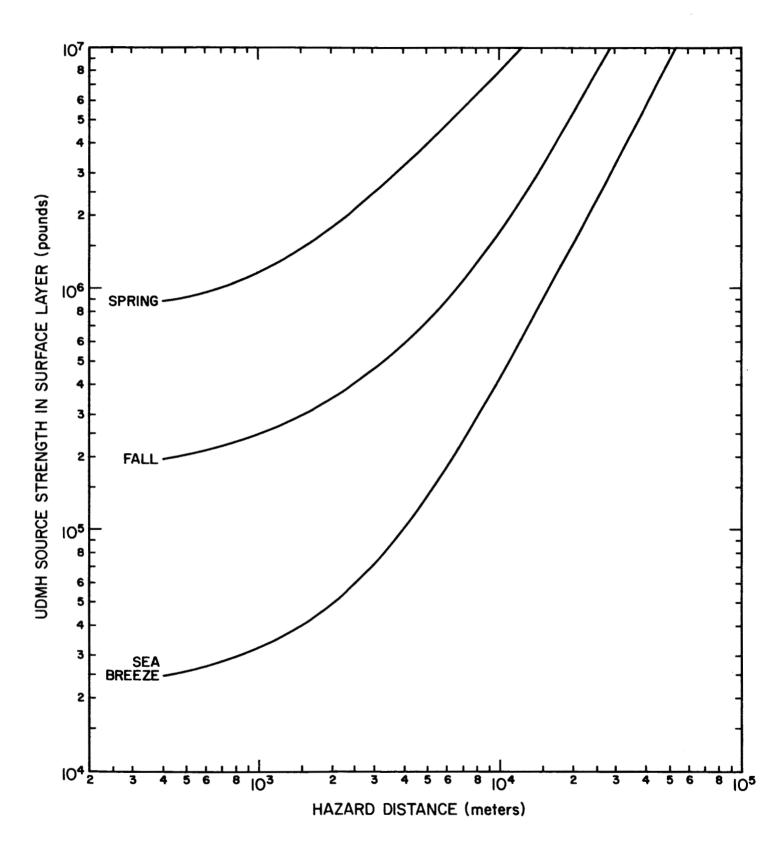


FIGURE 6-20. Hazard distances for a peak UDMH concentration of 100 ppm for various surface-layer source strengths.

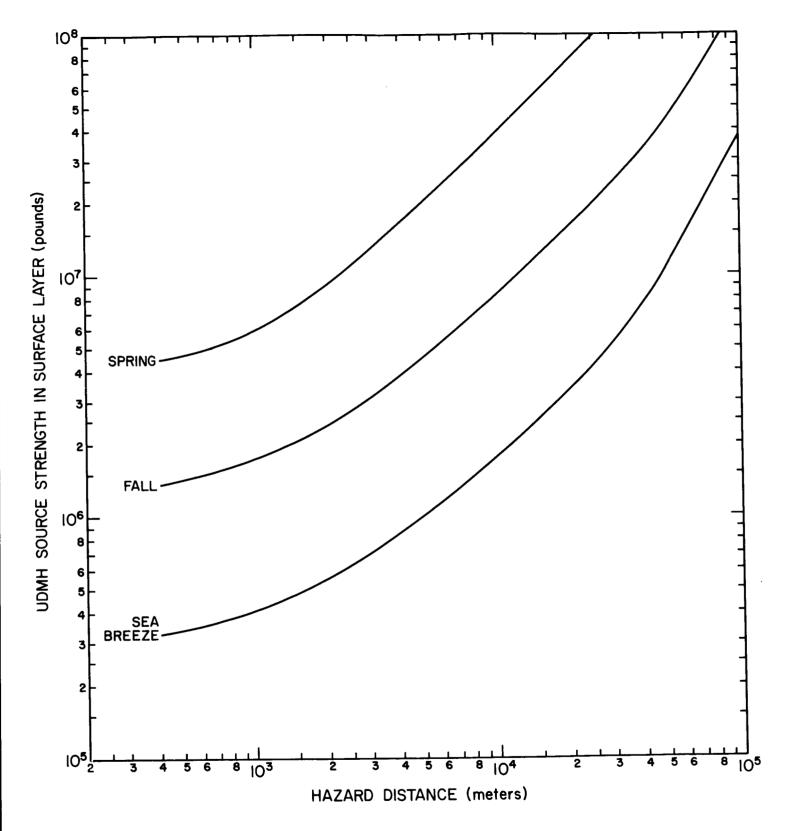


FIGURE 6-21. Hazard distances for a 10-minute average UDMH concentration of 100 ppm for various surface-layer source strengths.

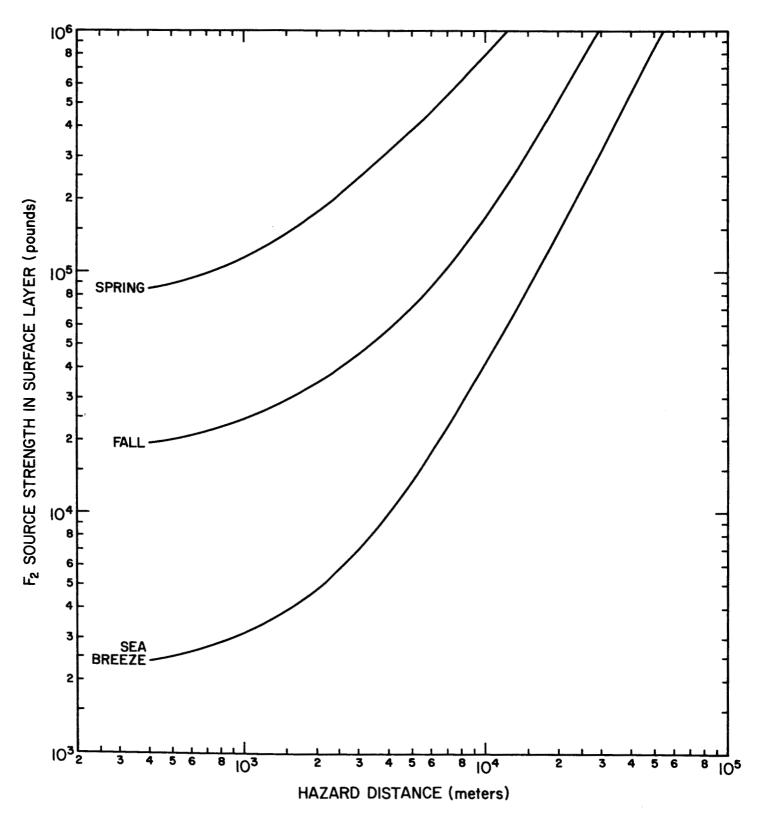


FIGURE 6-22. Hazard distances for a peak  ${\bf F}_2$  concentration of 15 ppm for various surface-layer source strengths.

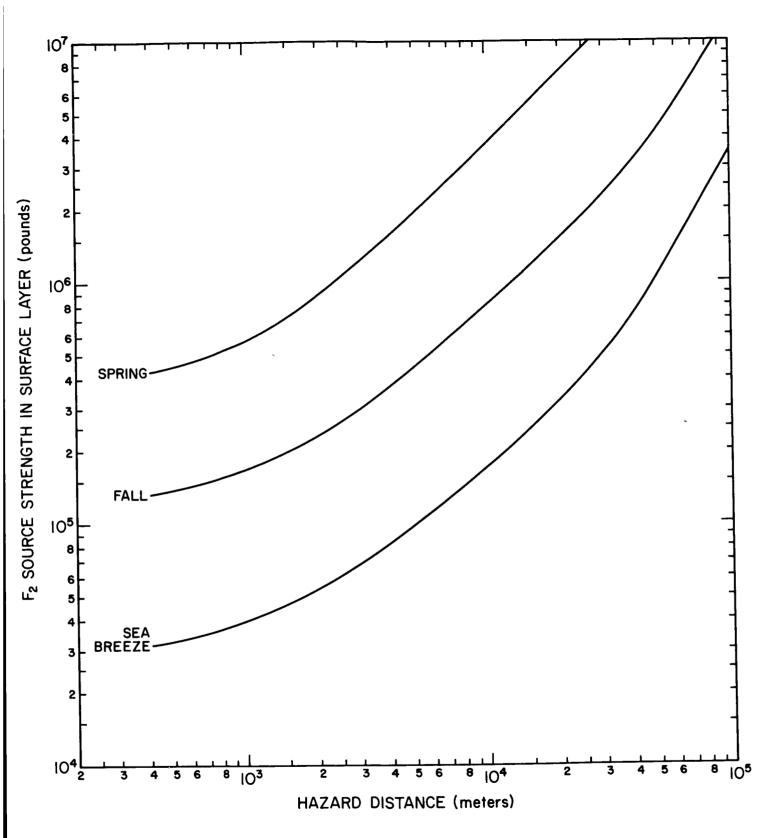


FIGURE 6-23. Hazard distances for a 10-minute average  ${\bf F}_2$  concentration of 15 ppm for various surface-layer source strengths.

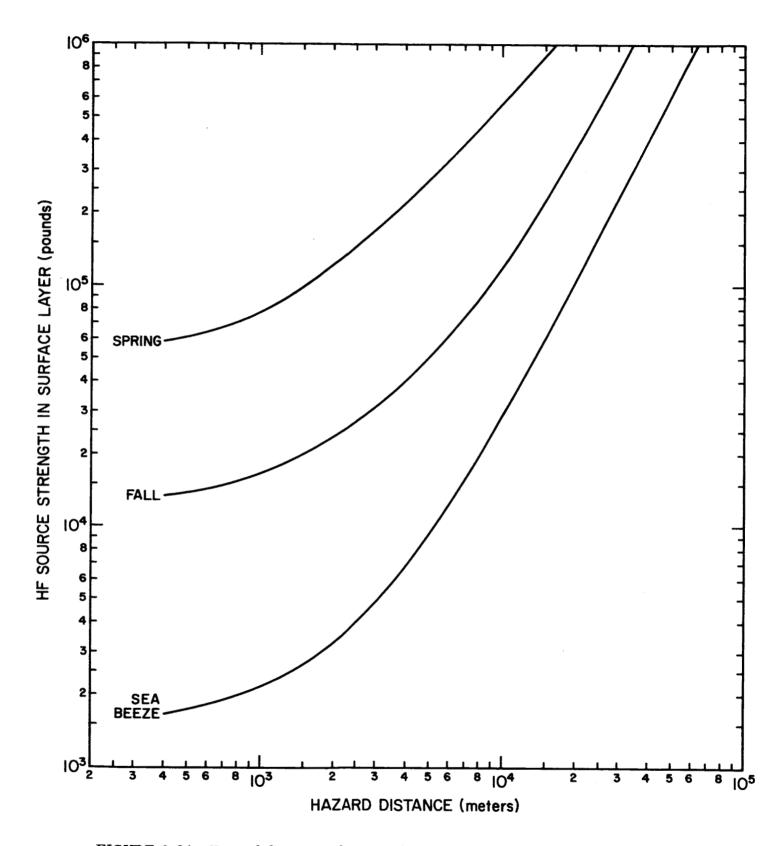


FIGURE 6-24. Hazard distances for a peak HF concentration of 20 ppm for various surface-layer source strengths.

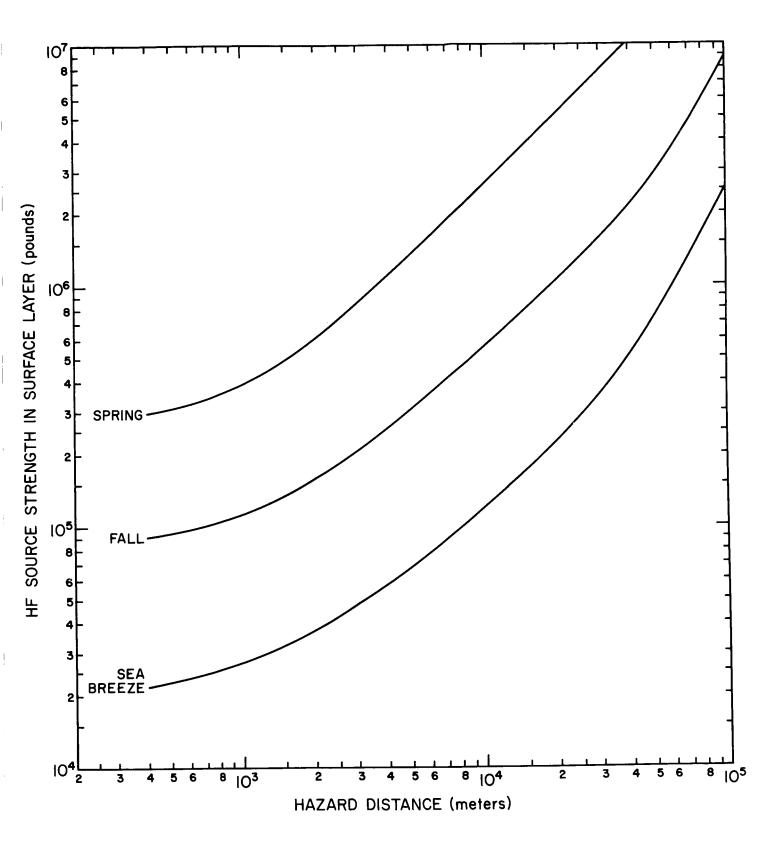


FIGURE 6-25. Hazard distances for a 10-minute average HF concentration of 20 ppm for various surface-layer source strengths.

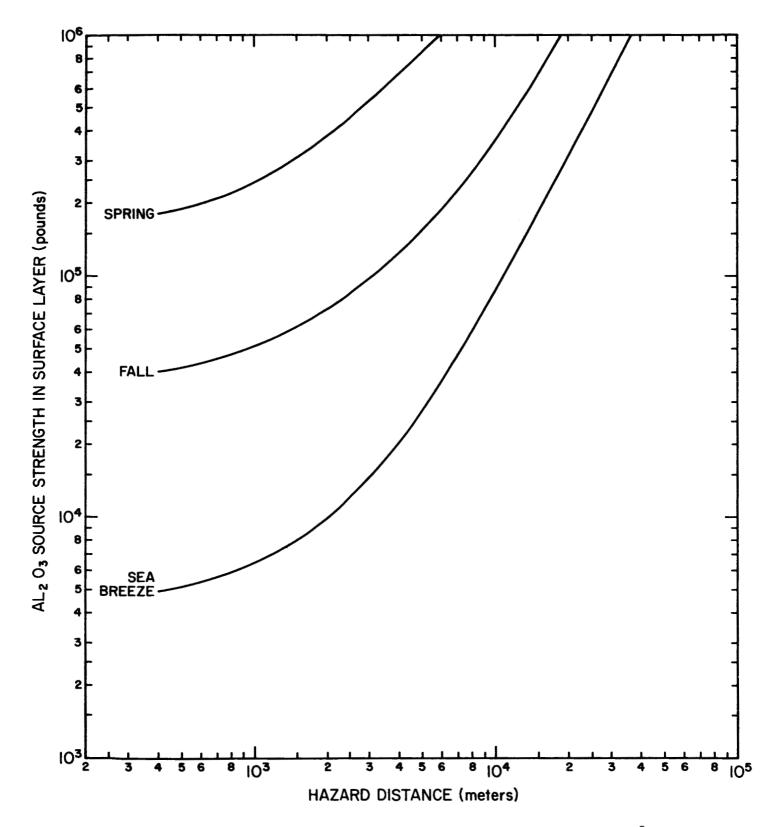


FIGURE 6-26. Hazard distances for a peak  ${\rm Al_2O_3}$  concentration of 50 mg m  $^{-3}$  for various surface-layer source strengths.

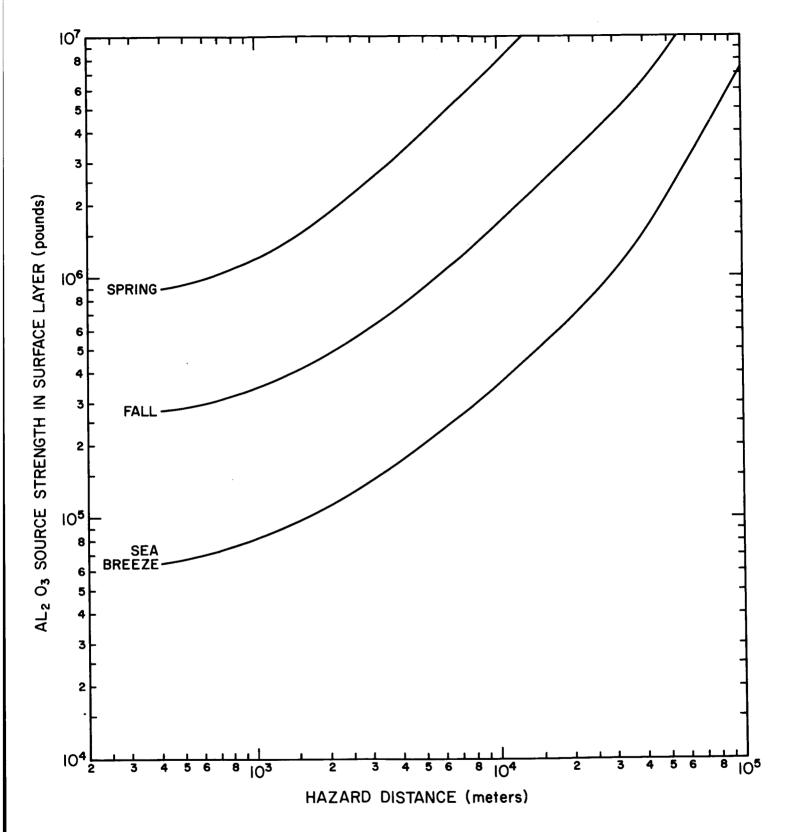


FIGURE 6-27. Hazard distances for a 10-minute average  ${\rm Al_2O_3}$  concentration of 50 mg m<sup>-3</sup> for various surface-layer source strengths.

#### REFERENCES

- Briggs, G. A., 1969: <u>Plume Rise</u>. TID 25075, Clearinghouse for Federal Scientific and Technical Information, Springfield, Va., 22151, 80.
- Briggs, G. A. 1970: Some recent analyses of plume rise observation. <u>Paper ME-8E</u> presented at the Second International Clean Air Congress of the Int. Union of Air Poll. Prevention, Dec. 6-11, 1970, Washington, D. C.
- Dumbauld, R. K., et al., 1970: Handbook for estimating toxic fuel hazards. Final Report under Contract No. NAS8-21453, NASA Report CR-61326, Marshall Space Flight Center, Alabama.
- Dumbauld, R. K., 1971: Review of cloud rise problem. Technical Note submitted to NASA, Marshall Space Flight Center, Alabama, under Contract No. NAS8-26673.
- Record, F. A., et al., 1970: Analysis of lower atmospheric data for diffusion studies. Final Report under Contract No. NAS8-30503, NASA Report CR-61327, Marshall Space Flight Center, Alabama.
- Smith, J. W. and W. W. Vaughan, 1961: Monthly and annual wind distribution as a function of altitude for Patrick Air Force Base, Cape Canaveral, Florida, NASA Technical Note D-610, George C. Marshall Space Flight Center, Huntsville, Alabama.
- Susko, M., J. W. Kaufman and K. Hill, 1968: Plume rise and growth of static test vehicle engine exhaust clouds. NASA TM X-53782, Aero-Astrodynamics Research Review No. 7, George C. Marshall Space Flight Center, NASA Marshall Space Flight Center, Alabama, 146-168.
- Thayer, S. D., M. W. Chandler, and R. T. Chu, 1970: Rise and growth of space vehicle engine exhaust and associated diffusion models. GEOMET, Inc. Final Report to NASA under Contract NAS8-24438, NASA CR-61331, NASA-George C. Marshall Space Flight Center, Alabama, 187.

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